

AMENDMENTS TO THE SPECIFICATION

Please replace Paragraph [0040] with the following paragraph rewritten in amendment format:

[0040] Referring to Figure 1, an exhaust flow nozzle 10 is shown in accordance with a preferred embodiment of the present invention. The exhaust flow nozzle forms a tubular component having a continuous, linear tapering outer wall 13 over its entire length, an upstream end 12 and a downstream end 14. A jet engine 16 generates exhaust gases which are directed into the nozzle 10 and which exit through the downstream end 14. Downstream end 14 includes a beveled edge 16. Beveled edge 16 is beveled at an angle 19 of preferably of about 5° - 50° , although it will be appreciated that this angle could vary from this range to meet the needs of a specific aircraft application. The above-mentioned angle is measured relative to a plane orthogonal to the longitudinal center line 18 extending longitudinally through the nozzle 10. In effect, the downstream end 14 forms a "lip" or extension 20 which helps to alter the flow of exhaust gasses exiting from the nozzle 10 in a manner that reduces sound pressure levels at an area below the nozzle 10 when the nozzle is affixed to an aircraft.